SERVICE



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Shoul a Shoune
APR 1 1994

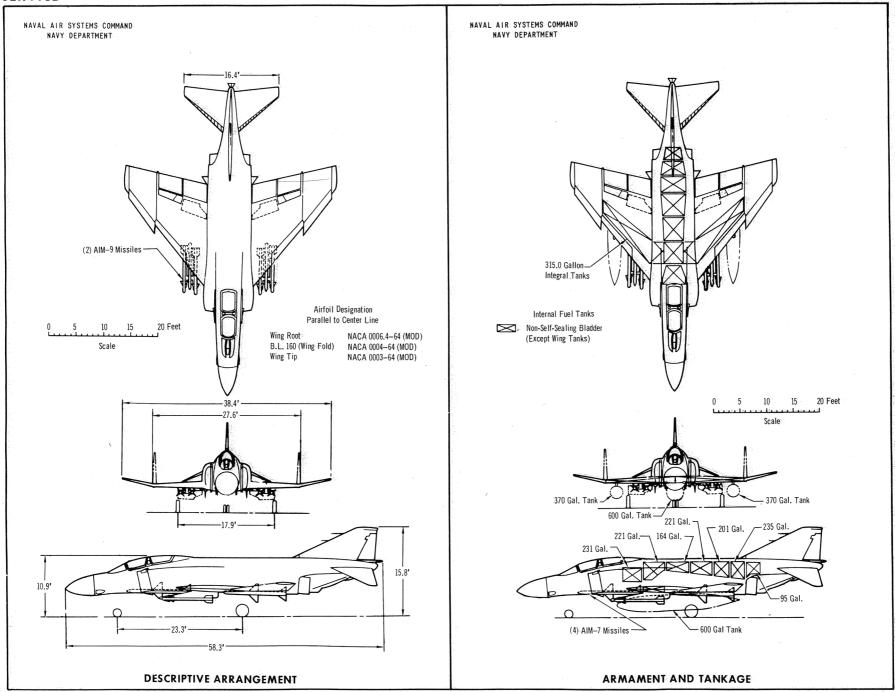
PUBLIC AFFAIRS OFFICE
NAVAL AIR SYSTEMS COMMAND

## STANDARD AIRCRAFT CHARACTERISTICS F-4S PHANTOM II

MCDONNELL

NOTE:

ALL INQUIRIES CONCERNING DATA IN THIS CHART SHOULD BE DIRECTED TO NAVAIR, CODE AIR-53012



POWER PL	ANT	ELECTRONI	CS (Cont)	WEIGHTS					
Number and Model Manufacturer Specifications: Type	(2) J79-GE-10B General Electric G.E. E-2039-A Axial Flow Turbojet	Air-to-Air IFF UHF Communication Radar Beacon Data Link	AN/APX-76A KY-28 AN/APN-154 AN/ASW-25B	Loading	Lb.	Subsonic L.F.	Superson L.F.		
Augmentation Length with A/B Diameter	Afterburner 208.69 Inches (Cold) 39.06 Inches (Cold)	MISSION AND	DESCRIPTION	Empty Basic Design	31,745 32,143 37,500	8.5	6.5		
Dry Weight Tail Pipe	3871 Lb. Variable Position	The F-4S is an F-4J modified by the install changes.	lation of maneuvering slats and other	Combat Maximum takeoff	42,640	7.5	5.7		
RATING	S*	It is a two-place, twin-jet, general purpose		Field Cat. Maximum Landing	56,000 56,000				
Static Thrust at Sea Level Power Setting (Lb.) Maximum (A/B) 17,820	RPM 97.1% 7,460	destruction of enemy aircraft. The capabil weapons enables the aircraft to perform in missions. Basic armament is four air-to-air under the fuselage plus up to four wing py diverse combination of conventional bomb	termediate and long range attack missiles carried semi-submerged Ion mounted air-to-air missiles. A ıs, rocket packages, gun pods, and	Field Arrest	46,000 a 38,000	at 10 ft/sec			
Military 11,810 Normal 11,100 90% Normal 10,000	97.1% 7,460 96.7% 7,435 92.9% 7,140	fuel tanks can be carried on five stations be external tanks plus a retractable probe for range missions. The F-4S features swept w	inflight refueling provides for extended	FU	EL AND	OIL			
75% Normal 8,330 Idle 400  *As defined in G.E. Spec. E-2039-A R: 3.4.4 Table 1 and subject to conditions with A/B and military is 30 minutes be below normal power is continuous.	89.8% 6,900 65.1% 5,000 ev, Dated 30 Aug 1974, Para. s therein. Maximum time	compression-ramp inlets, maneuvering slatt layer control. Lateral control is achieved b An all-movable stabilator provides longitud  The AWG-10A Missile Control Subsyste and control functions in the launching of a Weapons Subsystem and an All-Attitude B of conventional bombs and rockets.	s and trailing edge flaps with boundary y ailerons in combination with spoilers. Jinal control. em provides the necessary guidance pir-to-air missiles. A Multiple	No. Tanks 7 2 1	Gallor 1,368 630 600 740	Fus V Fus [	ation elage, Bladder Ving, Integral elage, External Drop Ving, External Drop		
ELECTRON	NICS	Other electronics includes a CNI system cations, navigation and identification subs		Grade Specification			or JP5 L-F-5624B-1		
CADC CNI Systems TACAN	A/A24G-33 AN/ARN-86	Computer (CADC) which senses air data p signals to various subsystems; the Automar which provides three axis stability augmen altitude hold, and the Radar Altimeter.	arameters and supplies appropriate tic Flight Control Subsystem (AFCS)	Integral with Engines Usable Tank Capacity p Specification MIL-L-23		5.1	5 Gal.		
ADF ECS 3500 Channel UHF Comm.	AN/ARA-50 LS-459/AIC RT-793A/ASQ	Equipment includes a pressurized cabin system, and anti-G non-pressure suit and fo		ORDNANCE					
IFF AFCS Navigational Computer Altimeter ' Missile Control System Includes Radar Missile Control Group	KY-532A/ASQ AN/ASA-32L AN/ASN-39A AN/APN-194 AWG-10A AN/APG-59 ( ) OA-6822	DEVELOR  Configuration Change Author First Flight Prototype First Flight Production		Four AIM-7 Missiles on I Air-to-Air Missiles AIM-7 AIM-9			board Outboa Jing Wing 2 4		
All-Attitude Bombing System Vertical Reference Set Approach Power Control Set	AN/AJB-7 AN/ASN-70 AN/ASN-54 (V)	DIMENS	SIONS	Conventional Bombs MK-81 LDGP		6	6 12		
Space Provisions for Installation of Data Link Visual Target Acquisition System ( ) Indicates Future Mod Identification	AN/ASW-25B AVG-8	Wing Area Span M.A.C. Sweepback (25% Choro)	530 Sq ft 38.4 ft 16.04 ft 45 <sup>0</sup> +1 <sup>0</sup>	MK-82 LDGP MK-83 LDGP MK-81 Snakeye MK-82 Snakeye MK-82 LGB MK-83 LGB Rockeye CBU		6 3 6 6 2 1 5	6 12 4 4 6 12 6 12 2 4 2 2 4 12		
The following equipment is effective 153851 and 155529 and up but was rdrag of the airplane for purposes of p	not included in the weight or	Incidence Dihedral - Inner Panel Outer Panel Length	0 <sup>0</sup> +12 <sup>0</sup> 58.3 ft	Rocket Packages LAU-10/A (19-2.75 in per Package) LAU-10/A (4-5.00 in		3	6 6		
Warning Decoy Dispenser Countermeasure Set	AN/ALR-45 AN/ALE-29A AN/ALQ-126	Height Wheelbase Tread MLG Tires	15.8 ft 23.3 ft 17.9 ft 20 x 11.5-14.5 Type VIII	Rockets per Package) Gun Pod MK-4 (20 mm)		3 1	6 6		
	-	NLG Tires	18 x 5.7 Type VII						

				PERFORM	ANCE SUMM	ARY [J]			
			1 Fighter	3 Fighter	5 Fighter	7 Area Intercept	9 Deck Launched Intercept	11 Ferry	13 HI-LO-HI Attack
TAKEOFF LOADING CONDITION  Take – off Weight Lb.			(4) AIM-7	(4) AIM-7 (1) 600 Gal. Tank (I)	(4) AIM-7 + (4) AIM-9 + (1) 600 Gal. Tank (1)	(4) AIM-7 + (1) 600 Gal, Tank (I)	(4) AIM-7 + (1) 600 Gal. Tank (1)	(2) 370 Gal. Tanks + (1) 600 Gal. Tank	(8) MK -82 S.E. + (2) 370 Gal. Tanks
			48,074	52,509	53,945	52,509	52,509	55,691	56,000
Fuel Internal/External (JP-5, 6.8 Lb/	Gal )	Lb.	13.587/-	13,587/4,080	13,587/4,080	13,587/4,080	13,587/4,080	12,941/9,112 M	13,587/3,711 N
Payload Missiles/Bombs	Jan. /	Lb.	1,820/-	1,820/-	2,608/-	1,820/	1,820/-	-/-	-/4,560
Wing Loading		Lb./Sq.Ft.	90.7	99.1	101.8	99.1	99.1	105.1	105.7
Stall Speed – (Approach Power)		Kn.	128	134	136	134	134	138	138
Takeoff Run at S.L. – Calm	A/B	Ft.	3,680/2,150	4,525/2,610	4.830/2.775	4,525/2,610	4,525/2,610	5,230/2,980	5,300/3,020
Takeoff Run at S.L. – 25 Kt. Wind	A/B	Ft.	2,980/1,630	3,720/2,020	3,980/2,160	3,720/2,020	3,720/2,020	4,340/2,360	4,400/2,400
Takeoff to Clear 50 Ft Calm	A/B	Ft.	5,020/3,300	6,000/3,860	6.340/4.050	6.000/3,860	6,000/3,860	6,775/4,310	6,850/4,350
Maximum Speed/Altitude	A A	Kn./Ft.	587/25,000	579/25,000	569/25,000	579/25,000	579/25,000	573/25,000	559/20,000
Rate of Climb at S.L.	A	fpm	10,700	9,200	8,400	9,700 L	28,500 B, L	8,300	7,150
Time: S.L. to 20,000 Ft.	A/B	L Min.	2.68/0.78	3.07/0.85	3.60/0.92	4.07/1.85 K	-/1.85 K	3.53/0.96	4.68/1.16
Time: S.L. to 30,000 Ft.	A/B	L Min.	4.65/1.38	5.63/1.52	6.88/1.67	6.63/2.52 K	-/2.52 K	6.80/1.67	10.3/2.11
Service Ceiling (100 fpm)	A	Ft.	39,650	37,200	35,650	38,200 L	47,700 B, L	35,350	31,300
Combat Range		Na.Mi.	871	1,100	994			1,490/1,347 C	790
		Kn.	485	485	484			485/483	472
Average Cruising Speed			35,677		34,956			34,300/34,300	31,100
Cruising Altitude Initial		Ft.		35,063	37,791			41,800/41,350	37,750
Cruising Altitude Final		Ft.	38,175	38,075			149	41,000/41,330	322
Combat radius		Na.Mi.	279	437	416	391			
Mission Time	D	Hr.	1.18	1.83	1.75	1.59	0.52	_	1.45
Average Cruising Speed		Kn.	487	487	486	485	485	_	479
C.A.P. Loiter Time	E	Hr.	0.65	1.28	1.05	_	_	_	
Mission Time	D	Hr.	1.32	1.96	1.75	-		-	
						(8)	(10)	(12) Tanka	<u>[14]</u>
COMBAT LOADING COM	IDITION		(4) AIM-7	(4) AIM-7	(4) AIM-7 + (4) AIM-9	(4) AIM-7	(4) AIM-7	Dropped	(2) TER at B.L. 81.50 + (1) MER at <b>©</b>
Combat Weight		Lb.	42,640	45,139	46,575	45,139	45,139	45,659	43,906
Engine Thrust			MAXIMUM	MILITARY	MAXIMUM	MAXIMUM	MAXIMUM	MAXIMUM	MILITARY
Fuel	F	Lb.	8,153	10,600	10,600	10,600	10,600	12,941	10,379
Combat Speed / Combat Altitude		Kn./Ft.	830/10,000	612/10,000	757/10,000	821/50,000	1,111/36,089	1,117/41,800	600/S.L.
Rate of Climb at Combat Altitude	G/H	fpm	-/29,200	-/10,013	-/25,400	80/60	9,930/9,050	6,340/5,300	-/10,810
Combat Ceiling (500 fpm)	G/H	Ft.	50,500/50,500	-/39,780	45,800/48,420	49,450/49,400	49,450/49,400	49,850/49,400	-/38,680
Rate of Climb at S.L.		fpm	35,200	11,480	30,800	33,250	33,250	33,700	10,810
Maximum Speed at S.L.		Kn.	705	616	678	705	705	713	600
Maximum Speed at S.L.		Kn.	1,113	589	1.022	1,111	1,111	1,124	576
					36,089	36,089	36,089	36,089	25,000
Altitude		Ft.	36,089	25,000	30,009	30,009	30,003	30,003	23,000
Landing Weight	-	Lb.	36,921	37,178	38,713	37,178	37,178	35,495	36,179
Fuel		Lb.	2,434	2,640	2,739	2,640	2,640	2,777	2,653
		Kn.	122	122	124	122	122	119	120
			<u> </u>	113	116	113	113	111	112
Stall Speed-Power-Off		Kn							,
Stall Speed-Power-Off Stall Speed-Approach Power	Angeles and the second	Kn.	113				2 990		2 920
Stall Speed-Power-Off		Kn. Ft. Ft.	2,975 4,820	2,990 4,890	3,100 5,000	2,990 4,890	2,990 4,890	2,875 4,780	2,920 4,825

## Notes:

A. Military thrust

В.

Maximum thrust

Ferry range with tanks dropped/retained €.

Mission time excludes warm-up, takeoff and reserve time H. Subsonic climb speed schedule D.

E. C.A.P. radius = 150 nm

60% of takeoff fuel or full internal fuel, whichever is less J.

G. Supersonic climb speed schedule

I. External fuel tanks dropped when they become empty L. No 5% service tolerance on fuel flow rate

K. Includes time increment due to ground operation and M. acceleration to climb speed

Includes weight reduction due to ground operation and fuel consumed during climb

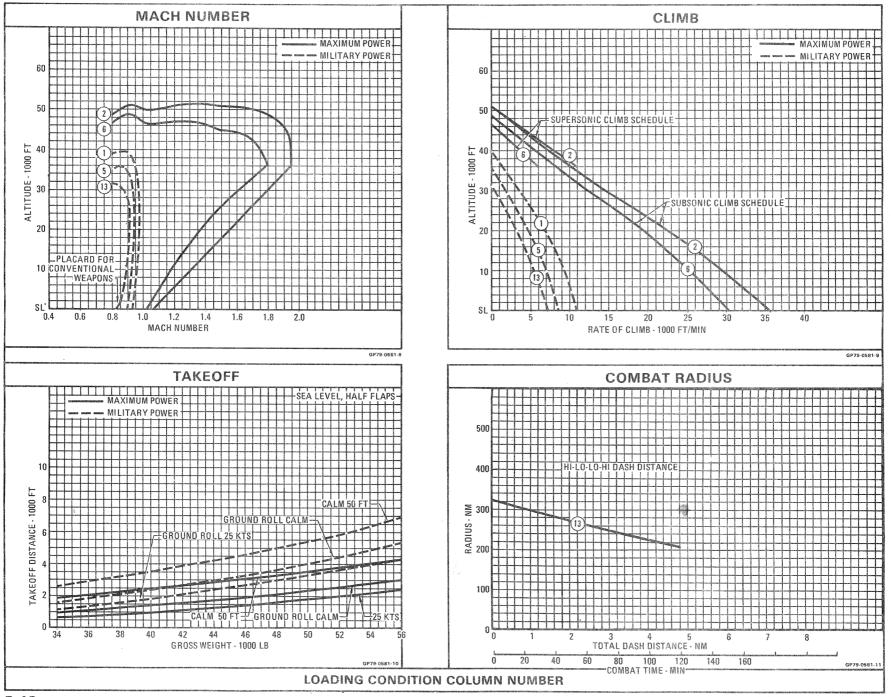
No fuel in tank no. 7 to maintain CG limits N. External fuel off-loaded. Weight limitation

MISSION SUMMARY-ALTERNATE LOADINGS [A]												
				CLOSE SUPPORT		LO-HI	HI-HI-HI  5 MIN. @ MILITARY THRUST (BEST CRUISE ALTITUDE)		ro-ro-ro		HI-LO-HI	
			1 HOUR LOITER @5000 FEET		SEA LEVEL 100 NA. MI.						5 MIN. MILITAR VMAX LEVEL	Y THRUST
EXTERNAL STORE LOADIN	G	T.O.G.W. LB.	COMBAT RADIUS NA. MI.	MISSION TIME HOUR (B)	COMBAT RADIUS NA. MI.	MISSION TIME HOUR (B)	COMBAT RADIUS NA. MI.	MISSION TIME HOUR (B)	COMBAT RADIUS NA. MI.	MISSION TIME HOUR (B)	COMBAT RADIUS NA. MI.	MISSION TIME HOUR (B)
(8) MK-82 SE + (2) 370 GAL. TANKS	(C) (F)	56,000	150	1.64	226	1.27	419	1.85	202	1.4	322	1.45
(6) MK-82 SE + (2) 370 GAL. TANKS	(C) (D)	54,997	220	1.91	289	1.5	500	2.16	231	1.55	392	1.71
(6) MK-82 SE + (4) AIM-7 + (2) 370 GAL. TANKS	(C) (E) (F)	56,000	161	1.67	236	1.28	437	1.9	208	1.39	335	1.48
(24) MK-81 SE + (4) AIM-7	(D) (E) (G)	56,000	_	_	<b>.</b>	_	197	0.96	103	0.77	124	0.64
(5) MK-83 SE + (2) 370 GAL. TANKS	(C) (F)	56,000	154	1.65	228	1.26	429	1.87	203	1.39	327	1.45
(8) MK-83 + (4) AIM-7	(D) (E) (G)	56,000	11 (H)	1.06	105 (H)	0.76	227	1.04	110	0.8	138	0.67
(2) MK-82 SE+ (2) 370 GAL. TANKS (1) 600 GAL. TANK	+ (C) (D) (F)	56,000	323	2.35	385	1.9	596	2.56	281	1.87	487	2.11
(6) MK-82 SE + (4) AI (4) AIM-9 + (1) 600 GAL. TANK	M-7 + (C) (E) (I)	56,000	90 (J)	1.44	173 (J)	1.08	342	1.53	170	1.18	250	1.14

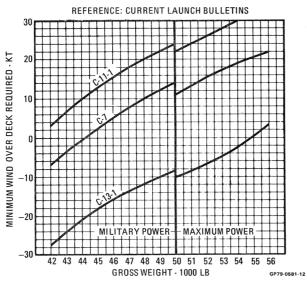
#### NOTES:

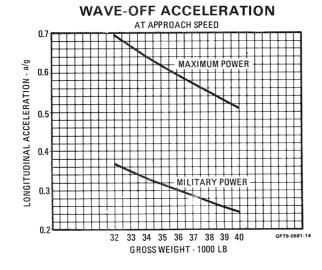
- A. NO 5% SERVICE TOLERANCE ON FUEL FLOW RATES.
- B. TIME EXCLUDES WARM-UP, TAKEOFF AND LANDING TIMES.
- C. EXTERNAL FUEL TANKS DROPPED WHEN THEY BECOME EMPTY.
- D. NO FUEL IN TANK NO. 7 TO REMAIN WITHIN C.G. LIMITS
- E. MISSILES RETAINED

- F. EXTERNAL WING TANK FUEL OFF-LOADED TO REMAIN WITHIN WEIGHT LIMITATION.
- G. INTERNAL FUEL OFF-LOADED TO REMAIN WITHIN WEIGHT LIMITATION.
- H. CRUISE ALTITUDE = 5,000 FT.
- I. CENTERLINE TANK FUEL OFF-LOADED TO REMAIN WITHIN WEIGHT LIMITATION.
- J. CRUISE ALTITUDE = 20,000 FT.

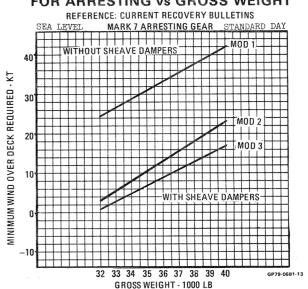


## MINIMUM WIND OVER DECK REQUIRED FOR CATUPULTING vs GROSS WEIGHT

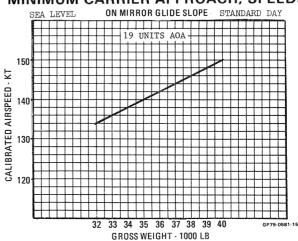


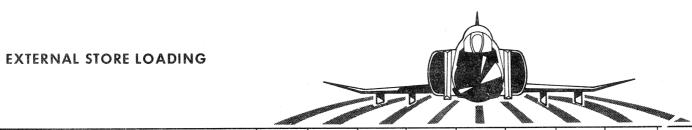


## MINIMUM WIND OVER DECK REQUIRED FOR ARRESTING vs GROSS WEIGHT



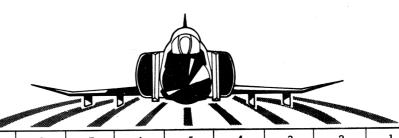
## MINIMUM CARRIER APPROACH, SPEEDS





	Allegar	400000000	Allen		****				-
STORE	9	8 .	7	6	5	4	3	2	1
GUIDED MISSILES - AIR-TO-AIR AIM-7 AIM-9		1 2	1	1		1	1	1 2	,
ROCKET LAUNCHERS  LAU-3A, 60/A, -61/A-61A/A  LAU-69A, -69A/A  LAU-10A, -10A/A, -10B/A  LAU-32B/A, -56/A, -68/A, -68B/A  LAU-33A/A	2 2 3 3	3 3 3 3			3 3 3 3		1	3 3 3 3	2 2 3 3
CONVENTIONAL BOMBS  MK-81 250 LB LDGP  MK-82 500 LB LDGP  MK-83 1000 LB LDGP  MK-81 SNAKEYE  MK-82 SNAKEYE  MK-83 RETARD	6 6 2 6 6	3 3 2 3 3 2			6 6 3 6 6 2			3 3 2 3 3 2	6 6 2 6 6
GUIDED BOMBS MK-82 LGB MK-83 LGB	2	1 1			2 1			1 1	2 1
FIRE BOMBS MK-77 MOD4	3	2			3			2	3
CLUSTER BOMBS CBU-59/B ROCKEYE II	3 6	2 2			4 5			2 2	3 6
DISPENSERS AN/ALE-37	3	2	,	, .				2	3
FLARES/SENSORS MK-24, MK-45, MK-58 SUU-40/44 ADSID1	6 2 6	3			6			3	6 2 6
FUEL TANKS 370 GALLON 600 GALLON	1				1				1

## **EXTERNAL STORE LOADING**



STORE	9	8	7	6	5	4	3	2	1
PRACTICE SHAPES  MK-76  MK-86  MK-87  MK-88  MK-89  MK-106  MK-124	6 6 6 2 3 6	3 3 3 2 2 2 3 3			6 6 3 3 6			3 3 2 2 2 3 3	6 6 2 3 6
AQM-37A TGT/LAU-24A LAUNCH					1				
MK-12 SMOKE TANK	1	2						2	1
CTU-1 DELIVERY CONTAINER	1				1				1
LB-30A PHOTO POD	-					1			
NAMAR CAMERA POD		1						1	
ALQ-120 ECM POD		1				11		1	
MK-4 GUN POD					1	1.4.			
D-704 REFUELING (BUDDY) TANK		,			1				
RMU-8A TOW TARGET					1				,
ASDC CONTAINER	2	2						2	2
RCPP 105 STARTER POD					1				
CNU169/A DELIVERY CONTAINER		1				ą.		1	
AN/ALE-29 (INTERNAL)	,								

#### FIGHTER ESCORT

- 1. ENGINE START, TAXI, TAKEOFF AND ACCELERATE: 5 minutes normal static sea level thrust. (See Note)
- CLIMB: On course at military thrust from sea level to best cruise altitude (not to exceed cruise ceiling).
- CRUISE OUT: Cruise climb at best cruise altitude and MN (not to exceed cruise ceiling, drop external fuel tanks when empty).
- DESCENT: To 10,000 Ft. No fuel used or distance gained.
- COMEAT FUEL ALLOWANCE: 2 minutes with maximum thrust at MN 1.0 10,000 feet missiles retained. (No distance gained)
- CLIMB: On course at military thrust from 10,000 Ft. to best cruise altitude (not to exceed cruise ceiling).
- CRUISE BACK: Cruise climb at best cruise altitude and MN (not to exceed cruise ceiling).
- DESCENT: To sea level. No fuel used or distance gained.
- RESERVE: Fuel shall be 5% of total initial fuel plus fuel required for 20 minutes loiter at sea level at speeds for maximum endurance with both engines operating.

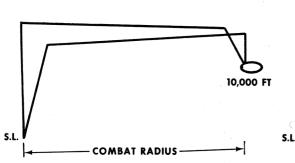
## **NOTES**

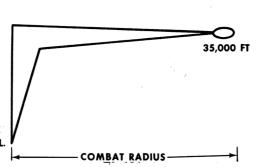
#### COMBAT AIR PATROL

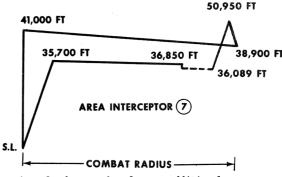
- ENGINE START, TAXI, TAKEOFF AND ACCELERATE: 5 minutes normal static sea level thrust. (See Note)
- CLIMB: On course at military thrust from sea level to best cruise altitude (not to exceed cruise ceiling).
- CRUISE OUT: Cruise climb at best cruise altitude and MN (not to exceed cruise ceiling drop external fuel tanks when empty).
- 4. LOITER: At maximum endurance speed at 35,000 feet. (No distance gained).
- ACCELERATE: At 35,000 feet at maximum endurance speed to MN 1.35 at maximum thrust. (No distance gained).
- COMBAT FUEL ALLOWANCE: 2 minutes with maximum thrust at MN 1.35 at 35,000 feet missiles retained. (No distance gained).
- CRUISE BACK: Cruise climb at best cruise altitude and MN (not exceed cruise ceiling).
- DESCENT: To sea level. No fuel used or distance gained.
- RESERVE: Fuel shall be 5% of total initial fuel plus fuel required for 20 minutes loiter at sea level at speeds for maximum endurance with both engines operating.

#### AREA INTERCEPTOR

- ENGINE START, TAXI, TAKEOFF AND ACCELERATE 5 minutes normal static sea level thrust. (See Note)
- CLIMB: On course at military thrust from sea level to best cruise altitude (not to exceed cruise ceiling).
- CRUISE OUT: At altitudes and speed for maximum range.
- ACCELERATE: At 36,089 feet to maximum supersonic climb speed.
- 5. CLIMB: On course with maximum thrust to supersonic combat ceiling.
- COMBAT FUEL ALLOWANCE: 2 minutes with maximum thrust at 1.5 MN at 50,000 Ft. (No distance gained).
- CRUISE BACK: At altitudes and speeds for maximum range.
- DESCENT: To sea level. No fuel used or distance gained.
- RESERVE: Fuel shall be 5% of total initial fuel plus fuel required for 20 minutes loiter at sea level at speeds for maximum endurance with both engines operating.







Notes: All takeoff gross weights of 52,000 pounds and over require afterburner therefore an additional 1 minute CRT in the takeoff allowance is added.

Fuel tanks dropped when empty.

Performance basis: Calculated data based on flight test of F-4E/S aircraft. No 5% service tolerance on fuel flow.

O LOADING CONDITION COLUMN NUMBER

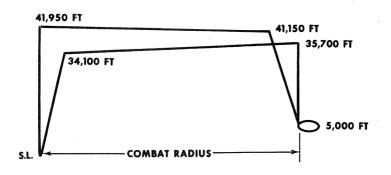
### **NOTES**

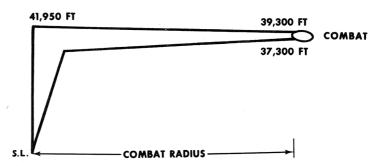
#### CLOSE SUPPORT

- ENGINE START, TAXI, TAKEOFF AND ACCELERATE: 5 minutes normal static sea level thrust. (See Note)
- CLIMB: On course with military thrust to optimum cruise altitude.
- CRUISE OUT: At altitudes and speeds for maximum range.
- DESCENT: To 5000 feet. No fuel, time or distance credited.
- LOITER: At 5000 feet for 1 hour at speeds and power for maximum endurance. Stores are dropped after loiter.
- 6. CLIMB: On course with military thrust to optimum cruise altitude.
- CRUISE BACK: At altitudes and speeds for maximum range.
- DESCENT: To sea level. No fuel, time or distance credited.
- RESERVE: Fuel shall be 5% of total initial fuel plus fuel required for 20 minutes loiter at sea level at speeds for maximum endurance with both engines operating.

#### HIGH ALTITUDE SUBSONIC: HI-HI-HI

- 1. ENGINE START, TAXI, TAKEOFF AND ACCELERATE: 5 minutes normal static sea level thrust. (See Note)
- CLIMB: On course with military thrust to optimum cruise altitude.
- 3. CRUISE OUT: At altitudes and speeds for maximum range.
- COMBAT FUEL ALLOWANCE: 5 minutes at maximum speed with military thrust (No distance credited). Drop stores after combat.
- 5. CRUISE BACK: At altitudes and speeds for maximum range.
- 6. DESCENT: To sea level. No fuel, time or distance credited.
- RESERVE: Fuel shall be 5% of total initial fuel plus fuel required for 20 minutes loiter at sea level at speeds for maximum endurance with both engines operating.





Notes: Configuration illustrated is (6) MK-82 S.E. + (2) 370 gallon tanks.

All takeoff gross weights of 52,000 pounds and over require afterburner therefore an additional one minute CRT in the takeoff fuel allowance is added.

Fuel tanks dropped when empty.

Performance basis: calculated data based on F-4E/S aircraft.

No 5% service tolerance on fuel flow.

LOADING CONDITION COLUMN NUMBER

### **NOTES**

#### HI-LO-HI

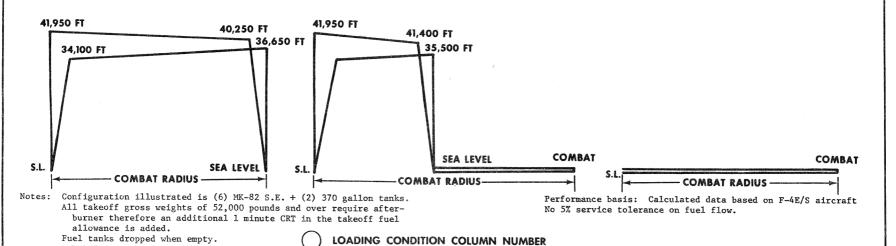
- ENGINE START, TAXI, TAKEOFF AND ACCELERATE: 5 minutes normal static sea level thrust. (See Note)
- CLIMB: On course with military thrust to optimum cruise altitude.
- CRUISE OUT: At optimum cruise altitude at speeds for maximum range.
- 4. DESCENT: To sea level. No fuel, time or distance credited.
- COMBAT FUEL ALLOWANCE: 5 minutes with maximum speed with military thrust (No distance credited). Drop stores after combat.
- 6. CLIMB: On course with military thrust to optimum cruise altitude.
- CRUISE BACK: At altitudes and speeds for maximum range.
- DESCENT: To sea level. No fuel, time or distance credited.
- RESERVE: Fuel shall be 5% of total initial fuel plus fuel required for 20 minutes loiter at sea level at speed for maximum endurance with both engines operating.

#### SEA LEVEL RUN-IN (HI-LO-LO-HI)

- ENGINE START, TAXI, TAKEOFF AND ACCELERATE: 5 minutes normal static sea level thrust. (See Note)
- CLIMB: On course with military thrust to optimum cruise altitude.
- CRUISE OUT: At altitudes and speeds for maximum range.
- DESCENT: To sea level. No fuel, time or distance credited.
- CRUISE OUT: 100 nautical miles at speeds for maximum range at sea level.
- COMBAT FUEL ALLOWANCE: 5 minutes at maximum speed with military thrust. (No distance credited). Drop stores after combat.
- CRUISE BACK: 100 nautical miles at speeds for maximum range at sea level.
- 8. CLIMB: On course with military thrust to optimum cruise altitude.
- CRUISE BACK: At altitudes and speeds for maximum range.
- DESCENT: To sea level. No fuel, time or distance credited.
- 11. RESERVE: Fuel shall be 5% of total initial fuel plus fuel required for 20 minutes loiter at sea level at speed for maximum endurance with both engines operating.

#### SEA LEVEL LO-LO-LO

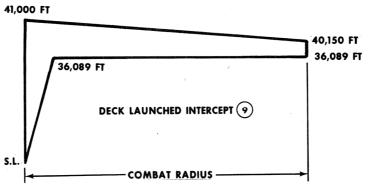
- ENGINE START, TAXI, TAKEOFF AND ACCELERATE: 5 minutes normal static sea level thrust. (See Note)
- CRUISE OUT: At sea level at speed for maximum range.
- COMBAT FUEL ALLOWANCE: 5 minutes at maximum speed with military thrust. (No distance credited). Drop stores after combat.
- CRUISE BACK: At sea level at speeds for maximum range.
- RESERVE: Fuel shall be 5% of total initial fuel plus fuel for 20 minutes loiter at sea level at speeds for maximum endurance with both engines operating.



## NOTES

#### DECK LAUNCHED INTERCEPT MISSION

- ENGINE START, TAXI, TAKEOFF AND ACCELERATE: 5 minutes normal static sea level thrust. (See Note)
- 2. CLIMB: On course to 36,089 feet with maximum thrust.
- 3. ACCELERATE: To 1.8 MN at 36,089 feet.
- 4. CRUISE OUT: On course at 36,089 feet at 1.8 MN with modulated afterburner.
- COMBAT FUEL ALLOWANCE: 2 minutes at 1.8 MN at 36,089 feet. (No distance credited)
- CRUISE BACK: At altitudes and speeds for maximum range.
- DESCENT: To sea level. No fuel, time and distance credited.
- 8. RESERVE: Fuel shall be 5% of total initial fuel plus fuel required for 20 minutes loiter at sea level at speeds for maximum endurance with both engines operating.



Notes: All takeoff gross weights of 52,000 pounds and over require afterburner therefore an additional one minute CRT in the takeoff fuel allowances is added.

Fuel tanks dropped when empty.

Performance basis: calculated data based on flight test of F-4E/S aircraft.

No 5% service tolerance on fuel flow.

LOADING CONDITION COLUMN NUMBER

# UKCLASSFED

WCLASSFED